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Generation of Arbitrary Lagrangian-Eulerian
(ALE) velocities, based on monitor functions,
for the solution of compressible fluid equations.

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Abstract

A moving mesh method is outlined based on the use of monitor functions. The method is developed from a weak conservation principle. From this principle a conservation law for the mesh position is derived. Using the Helmholtz decomposition theorem, this conservation law can be converted into an elliptic equation for a mesh velocity potential.

The moving mesh method is discretised using standard finite elements. Once the mesh velocities are obtained an Arbitrary Lagrangian Eulerian (ALE) [3] fluid solver is used to update the solution on the adaptive mesh.

Results are shown for the compressible Euler equations of gas dynamics in one and two spatial dimensions. Two monitor functions are used, the fluid density (which corresponds to a Lagrangian description), and a function which includes the density gradient. A variety of test problems are considered.

1 INTRODUCTION

In this paper an adaptive method is presented for the solution of multi-dimensional hyperbolic conservation laws of the form

$$U_t + r \cdot \nabla F(U) = 0 \quad \text{in } \Omega(t) \in [0; T]. \quad (1)$$

Here $U = U(x; t)$ is some m -vector of conserved variables and x and t are spatial and temporal variables respectively. The system (1) is solved in some spatial region $\Omega(t)$ having a boundary $\partial\Omega(t)$ which may or may not be moving in time.

2 FORMULATION OF THE MOVING MESH METHOD

Let $M(x; t) > 0$ be a user-defined monitor function which reflects the characteristics of the solution to (1). The mesh positions are chosen to satisfy

$$\int_{\Omega(t)} M \, d\Omega = \text{constant in time}; \quad (2)$$

for all $\Omega(t)$. In practice M will depend on the solution of the PDE (1) and its partial derivatives. Differentiating (2) with respect to time gives

$$\frac{d}{dt} \int_{\Omega(t)} M \, d\Omega = 0$$

$$\dot{x} = q + r\tilde{A} \quad (6)$$

Using (6) in equation (4) results in an elliptic equation for the velocity potential $\tilde{A}$ of the pseudo-fluid.

$$\int_{\Omega(t)} r \nabla (M r \tilde{A}) d\Omega = \int_{\Omega(t)} M_t d\Omega - \int_{\Omega(t)} r \nabla (M q) d\Omega \quad (7)$$

A boundary condition involving either $\tilde{A}$ or $\nabla \tilde{A} \cdot n$ is required.

We denote the finite element approximations to U , $\dot{x}$, v and $\tilde{A}$ by $\tilde{U}$, $\dot{X}$, V and Ψ respectively. The domain $\Omega(t)$ is partitioned into nonoverlapping computational cells and a patch of such cells surrounding the i^{th} node will be denoted by $\Omega_i(t)$. The test function w becomes one of the finite element basis functions w_i which form a partition of unity. All the finite element approximations are expanded in terms of these basis functions. In all the work presented we will take the basis functions to be piecewise linear functions, so hat functions in 1D and pyramid functions defined on triangular elements in 2D. Therefore the velocity potential equation (9) becomes

$$\int_{\Omega_i(t)} w_i r \nabla \cdot (M r \Psi) d\Omega = \int_{\Omega_i(t)} w_i M_t d\Omega - \int_{\Omega_i(t)} w_i r \nabla \cdot (M V) d\Omega; \quad (10)$$

The velocity potential equation (10) leads to a weighted stiffness matrix system for Ψ . The velocity is then recovered from the velocity potential using the weak form

$$\int_{\Omega_i(t)} w_i \dot{X} d\Omega = \int_{\Omega_i(t)} w_i r \Psi d\Omega; \quad (11)$$

which is equivalent to

$$\min_{\dot{X}_i} \|\dot{X}_i - r \Psi\|_{L_2}^2; \quad (12)$$

Equation (12) leads a set of mass matrix systems, one for each component of the velocity field $\dot{X}$. Once the velocity has been found we can use it in an ALE fluid solver and time-step the mesh.

4 NUMERICAL RESULTS

In this section, numerical results are presented for the solution of the two-dimensional compressible Euler equations. The Euler equations, along with the ideal gas equation of state, are solved on a moving adaptive mesh generated by the method outlined in the previous sections. In all the results shown the ratio of specific heats was taken to be $\gamma = 1.4$. Once a mesh velocity has been generated, by solving (10) and calculating the velocity through (12), the Euler equations need to be solved on a moving mesh. This is done by solving the ALE form of the Euler equations, which are the equations transformed into a moving frame of reference. In one spatial dimension this is done with the ALE form of the Roe Riemann solver [4, 5] and in two dimensions by

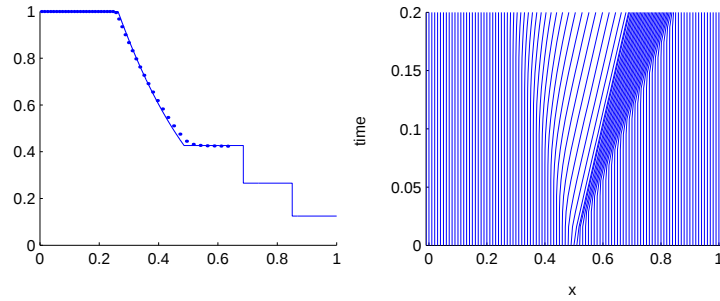


Figure 1: Lagrangian Solution to the 1D Sod shock tube problem at $t = 0.2$ using a Roe Riemann-solver with a superbee limiter on a moving mesh obtained from $M = \frac{1}{2}$. ($CFL = 0.5; N = 100$). Trajectories of the mesh points.

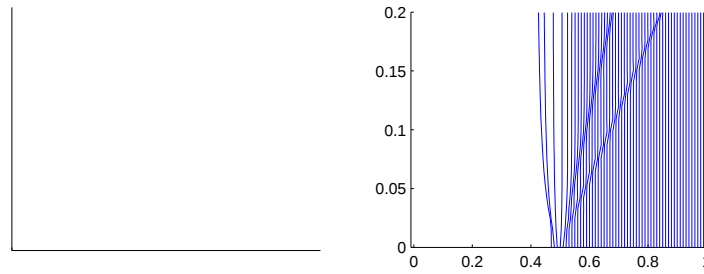


Figure 2: ALE Solution to the 1D Sod shock tube problem at $t = 0.2$ using a Roe Riemann-solver with a superbee limiter on a moving mesh obtained from $M = 1 + \frac{1}{2} \frac{\max_x j^{\frac{1}{2}} j}{j^{\frac{1}{2}} j}$, $\frac{1}{2} = \frac{1}{2}$. ($CFL = 0.5; N = 100; \bar{= 2}$). Trajectories of the mesh points.

the HLLC Riemann solver [7]. It should also be noted that in all the results shown we have taken the velocity field q to be zero.

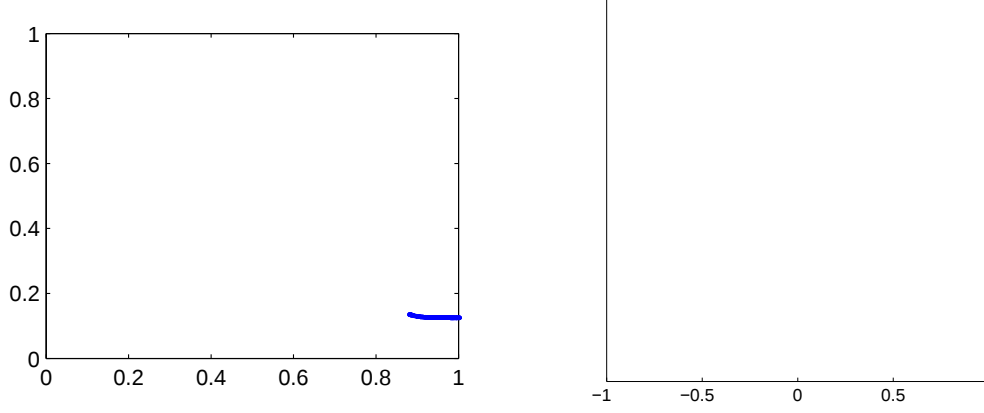


Figure 3: Density solution to the diverging cylindrical shock tube problem at $t = 0.2$ using a first order in space HLLC Riemann-solver on a moving mesh obtained from $M = \frac{1}{2}$. Mesh with 10201 nodes and 20000 triangles obtained at $t = 0.2$.

$M = 1 + \mathcal{O}(j^{\frac{1}{2}}_x j)$, with a suitable scaling $\mathcal{O}$. This monitor function was chosen so as to move mesh points into regions of the flow where there are large variations in the fluid density. Results for this problem are shown in figure 2. The trajectories clearly show how the mesh follows the velocity of the fluid in figure 1 and responds to the characteristics in figure 2. The results obtained from the moving mesh algorithm were compared to the exact solution, which was computed with an exact Riemann solver.

In 2D we solved a cylindrical shock problem. The problem consists of two regions of gas separated by a membrane at $x^2 + y^2 = \frac{1}{2}$. The gas at the centre of the region has a higher density and is at a higher pressure compared to the one outside the membrane.

This problem was also solved using two different monitor functions. The first monitor function was again chosen to used was $M = \frac{1}{2}$. Results for this monitor function can be seen in figure 3. The second monitor function used was $M = 1 + \mathcal{O}(j r^{\frac{1}{2}} j^2)$, where $\mathcal{O}$ was suitably chosen. Results for this monitor function can be seen in figure 4. The problem was solved numerically in 2D for each of these monitor functions and compared with a 1D radial computation of the problem computed on a very fine mesh. Although the density profiles are not strongly affected the mesh is clearly moving in a rational way. It is expected that considerable improvement will occur when a second order solver is implemented.

Figure 4: Density solution to the diverging cylindrical shock problem at $t = 0.2$ using a first order in space HLLC Riemann-solver on a moving mesh obtained from $M = 1 + \frac{1}{2}r^2$. Mesh with 10201 nodes and 20000 triangles obtained at $t = 0.2$.

5 CONCLUSIONS

A method for generating mesh velocities using monitor functions has been presented which can then be used in ALE fluid solvers. We have used the method outlined in sections 2 and 3 to produce an adaptive mesh for the solution of the Euler equations of gas dynamics in one and two spatial dimensions. Two test problems have been solved using different monitor functions leading to different types of mesh movement.

In future work we aim to use improved initial meshes for the problem, instead of equi-spaced meshes. We also need to increase the order of the HLLC Riemann solver in 2D in order to better resolve the density profile, which will in turn sharpen the adaptivity. Other monitor functions will be tried and work will also be done on investigating the influence of the rotational vector field q in (6) and how to choose it for a given problem.

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